

Systems of Particles + Center of Mass

General Principles

Relativity Principle

- Physics is the same in all inertial reference frames.
That is, all forces in Newton's law (in an inertial frame) have a physical origin.
- We can choose a convenient reference frame for solving a given physics problem.
- We want to understand the relationship between a lab frame + the rest frame of the center of mass (CM frame) for an object or group of particles.

- Procedure for change of one reference frame to another

- Work out change in positions, $y' \uparrow y \uparrow$

+ If origin of frame S is at

\vec{r}_2 according to frame S' ,

the position of a particle measured in the 2 frames

is related by

$$\vec{r}' = \vec{r} + \vec{r}_2$$

+ If S moves inertially w.r.t S' , $\vec{r}_2 = \vec{r}_0 + \vec{V}t$.

- Then you differentiate to find the relationship of velocities, combine to find KE, angular momentum, etc.

Center of Mass (CM) frame : frame where CM is at rest at the origin

- CM and momentum

- Consider a group of particles of masses m_i and positions \vec{r}_i in some lab frame S

+ The total mass is $M = \sum m_i$

+ The center of mass position is the mass-weighted average position

$$\vec{R} = \left(\sum_i m_i \vec{r}_i \right) / M$$

+ \vec{R} is the position of the origin of CM frame S^* relative to lab frame S (i.e., $\vec{r}^* = 0$)

+ For continuous materials, convert the sum to integrals

* Total Momentum

+ In lab frame, total momentum is

$$\vec{P} = \sum_i \vec{p}_i = \sum_i m_i \vec{v}_i = \frac{d}{dt} \left(\sum_i m_i \vec{r}_i \right) = M \vec{\dot{R}}$$

This is momentum of one mass M particle moving with CM position $\vec{R}(t)$. Call this "CM particle"

+ Residues

$$M \ddot{\vec{R}} = \vec{P} = \sum_i \vec{p}_i = \sum_i (\vec{F}_{ext,i} + \sum_j \vec{F}_{ij}) = \vec{F}_{ext}$$

where $\vec{F}_{ij} =$ force on particle i from particle j

and the double sum vanishes b/c $\vec{F}_{j,i} = -\vec{F}_{i,j}$ (Newton's 3rd law)

+ In other words, total momentum is conserved if there are no external forces + the CM motion is determined by the external forces

+ If the external force is uniform gravity,

$$\vec{F}_{ext,G} = m \vec{g} \Rightarrow \vec{F}_G = M \vec{g} = M \ddot{\vec{R}}$$

\Rightarrow Potential energy of uniform gravity is $V_G = -M \vec{g} \cdot \vec{R}$

- Energy of a group of particles

* Kinetic energy splits into CM contribution + CM frame

$$T = \frac{1}{2} \sum_i m_i \vec{v}_i^2 = \frac{1}{2} \sum_i m_i (\vec{v}_i^{**} + \vec{v}_C)^2$$

where

\vec{v}_C^{**} = position in CM frame

+ Simplify

$$T = \frac{1}{2} \sum_i m_i (\vec{r}_i^{*2} + \vec{p}_i^{*2} + 2\vec{r}_i^* \cdot \vec{p}_i)$$

$$= T^* + \frac{1}{2} M \vec{R}^{*2} + (\sum_i m_i \vec{r}_i^{*2}) \cdot \vec{R}$$

where T^* = KE measured in CM frame.

+ The last term = 0 b/c $\sum m_i \vec{r}_i^* = \vec{R}^* = \text{constant} (= 0)$

+ Together, KE is "CM particle" energy plus energy in CM frame

• 2-particle special case $T = \frac{1}{2} M \vec{R}^{*2} + T^*$

+ CM Frame KE is

$$T^* = \frac{1}{2} M_1 \vec{r}_1^{*2} + \frac{1}{2} M_2 \vec{r}_2^{*2}$$

+ Define relative position

$$\vec{r} = \vec{r}_1 - \vec{r}_2 \quad \text{If } \vec{r}_0 = \vec{r}_1 - \vec{r}_2 \text{ in lab frame}$$

+ B/c for CM frame

$$M_1 \vec{r}_1^* + M_2 \vec{r}_2^* = 0 \Rightarrow \vec{r}_1^* = \frac{M_2}{M} \vec{r}, \vec{r}_2^* = -\frac{M_1}{M} \vec{r}$$

+ Plug into KE + simplify

$$T^* = \frac{1}{2} m_1 \left(\frac{m_2}{M}\right)^2 \vec{r}^2 + \frac{1}{2} m_2 \left(\frac{m_1}{M}\right)^2 \vec{r}^2$$

$$= \frac{1}{2} \frac{m_1 m_2}{M} \vec{r}^2 = \frac{1}{2} M \vec{r}^2$$

where $M = m_1 m_2 / M = \text{reduced mass}$

+ Motion in CM frame is like motion of a single particle

b/c the second one "mirrors" the first.

• Potential Energy

+ Forces split into external forces (From outside the system) and forces between particles in the system

$$T = \frac{1}{2} \sum_i (p_i^2 + \vec{r}_i^2)$$

$$\text{Force on particle } i = \vec{F}_{i,\text{ext}} + \sum_{j \neq i} \vec{F}_{ij}$$

+ If the internal forces are conservative, then

they come from a potential. By 3rd law + relativity principle,
it must be central $V_{ij} = V_{ij}(\vec{r}_i - \vec{r}_j)$

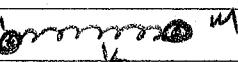
The total is an internal potential energy $V_{\text{int}} = \sum V_{ij}$

+ F

+ For rigid bodies, distances between particles don't change, so V_{int} = constant

+ We can work out $\frac{d}{dt}(T^* + V_{\text{int}}) = \sum_i \vec{r}_i^* \cdot \vec{F}_{i,\text{ext}}$

Forces $\propto m_i$ like uniform gravity or fictitious forces cancel

* Example: 2 masses connected by spring 

+ The spring has spring constant k and equilibrium length d . So $V_{\text{int}} = \frac{1}{2}k(r-d)^2$

+ Suppose these are moving in 2D, so there is also gravity. Then

$$L = T - V = \frac{1}{2}M\vec{R}^2 + \frac{1}{2}m\vec{r}_1^2 + \frac{1}{2}m\vec{r}_2^2 - \frac{1}{2}k(r-d)^2 + M\vec{g}\cdot\vec{R}$$

\Rightarrow

$$M\vec{R} = +M\vec{g}, \quad m\vec{r} = -k(r-d)$$

+ Notice how the equations separate.

- Angular Momentum

* Separation of CM contribution + CM frame value

+ We can re-write angular momentum as

$$\vec{j} = \sum_i m_i (\vec{r}_i^* + \vec{R}) \times (\vec{v}_i^* + \vec{V}) = \dots$$

$$= M\vec{R} \times \vec{V} + \sum m_i \vec{r}_i^* \times \vec{v}_i^* \equiv M\vec{R} \times \vec{V} + \vec{j}^*$$

+ This is angular momentum of CM and the angular momentum around the CM (\vec{j}^*)

+ For 2 particles, it again looks like one particle

$$\vec{j}^* = m_1 \vec{r}_1^* \times \vec{v}_1^* + m_2 \vec{r}_2^* \times \vec{v}_2^* = M\vec{r} \times \vec{v}$$

• Example: 2 masses on spring again moving on a frictionless surface:

+ If they slide differently, they rotate around each other in CM frame

+ Total lab angular momentum is

$$\vec{J} = M \vec{R} \times \frac{\vec{v}}{R} + m \vec{r} \times \vec{r}$$

The 2nd term is how much they rotate around each other

+ In cylindrical coords (ie, table in plane polar) for \vec{r} ,

$$\vec{J}^* = m r \hat{x} \dot{\hat{r}} = m r^2 \dot{\phi} \hat{\theta}$$

and

$$T^* = \frac{1}{2} m \dot{r}^2 + \frac{1}{2} m \dot{\phi}^2 + \frac{1}{2} m r^2 \dot{\phi}^2 = \frac{1}{2} m \dot{r}^2 + \frac{J^*^2}{2 m r^2}$$

like we saw last term.

Applications & Examples

- Two-Body Collisions

• With no external forces, total momentum $M \vec{R}$ is conserved.

+ Kinetic energy is $T = \frac{1}{2} M \vec{R}^2 + \frac{1}{2} m \vec{r}^2$

• + Since CM motion does not change, the max KE lost in an inelastic collision is $T^* = \frac{1}{2} m \vec{r}^2$

+ That's due to momentum conservation

• Elastic collision: KE conserved.

+ Again, $T = \frac{1}{2} M \vec{R}^2 + \frac{1}{2} m \vec{r}^2$ with $\frac{1}{2} M \vec{R}$ conserved
 $\Rightarrow \frac{1}{2} m \vec{r}^2$ is also conserved

+ This means the relative speed (magnitude of relative velocity \vec{r}) is the same before & after the collision

+ Example: In 1D, 2 objects collide with velocities $v_1 = 1 \text{ m/s}$, $v_2 = -2 \text{ m/s}$. They separate with velocities $u_1 = -2 \text{ m/s}$, $u_2 = 1 \text{ m/s}$ (Newton's Cradle)

- Gravity Assist: Space probe approaches a planet with relative speed u . Must leave w/ same relative speed.
 - + Planet is effectively infinitely massive, so CM frame is planet's rest frame.
 - + In solar frame, planet has constant speed v "horizontally". Probe initial velocity components are $(ucos\theta - v, usin\theta)$. To maintain same relative speed, final probe velocity is $(ucos\theta + v, usin\theta)$.
 - + Useful for sending ships far away b/c of "free" speed boost
 - + No need to do detailed orbit calculation

— Building Up a System

- We can group particles & build up KE & angular momentum in that way.
 - + Suppose we are in total CM frame of 2 objects, which we take as being made of many particles
 - $KE \rightarrow T = T_1 + T_2$, Ang. Mom is $\vec{J} = \vec{J}_1 + \vec{J}_2$
 - + But if the total mass + CM position of object 1 are M_1, \vec{R}_1 we see that
- $T_1 = \frac{1}{2} M_1 \vec{R}_1^2 + T_{1*}, \quad \vec{J}_1 = \frac{1}{2} M_1 \vec{R}_1 \times \vec{\omega}_1 + \vec{J}_{1*}$
- like before: same for object 2.
- + So you can think of groups of particles as an object and subdivide as much as you want

- Example: Two spinning hockey pucks
 - + 1st puck CM is located at \vec{R}_1 , and it is made of lots of individual atoms
 - + Then
- $T_1 = \frac{1}{2} \vec{R}_1 m \dot{\vec{v}}^2 = \frac{1}{2} M_1 \vec{R}_1^2 + T_{1*} = \frac{1}{2} M_1 \vec{R}_1^2 + \frac{1}{2} I_1 \vec{\omega}_1^2$
- and

$$\vec{J}_1 = M_1 \vec{R}_1 \times \vec{\omega}_1 + I_1 \vec{\omega}_1 \hat{z} \quad (\text{from what})$$

$\vec{R}_1 \hat{z} \vec{\omega}_1$ we know about rigid objects)

+ Repeat for 2nd hockey puck + combine

$$T = T_1 + T_2 = \left(\frac{1}{2} M_1 \vec{R}_1^2 + \frac{1}{2} M_2 \vec{R}_2^2 \right) + \frac{1}{2} I_1 \omega_1^2 + \frac{1}{2} I_2 \omega_2^2$$

$$= \frac{1}{2} M \vec{R}^2 + \frac{1}{2} M \vec{r}^2 + \frac{1}{2} I_1 \omega_1^2 + \frac{1}{2} I_2 \omega_2^2$$

where $M = \text{total mass}$, $\vec{R} = \text{total CM position}$, $\vec{r} = \vec{R}_1 - \vec{R}_2$

+ Also

$$\vec{J} = \vec{J}_1 + \vec{J}_2 = M \vec{R} \times \vec{R} + M \vec{r} \times \vec{r} + I_1 \hat{\omega}_1 + I_2 \hat{\omega}_2$$

• Similar logic can work for non-rigid objects

Example: Unwinding Rope

+ A rope of linear density μ wraps once around a disk of radius R . Disk rotates around center to unwind the rope

+ When disk is at angle $\theta = 2\pi$, it's all wound up.

Amount of hanging rope is $x = (2\pi - \theta)R$

+ Each piece of rope moves at speed \dot{x} as disk rotates with $\dot{\theta} = -\dot{x}/R$. Kinetic energy is

$$T = \frac{1}{2} \mu l \dot{x}^2 + \frac{1}{2} I \dot{\theta}^2 = \pi R^2 \mu \dot{x}^2 + \frac{1}{2} I \dot{x}^2 / R^2$$

where $l = 2\pi R = \text{length of rope}$.

+ If we set center of disk at $y = 0$, potential energy is given by CM of rope. This is located at

$$Y = \frac{1}{2\pi R \mu} \int dl \mu y = \frac{1}{2\pi R} \left[x \left(-\frac{x}{2} \right) + \int_0^{2\pi - x/R} d\theta R (2 \sin \theta) \right]$$

Given by CM of hanging part averaged w/ CM of wrapped part
Then

$$V = (2\pi R \mu) Y g = \mu g R^2 \left[1 - \cos \left(\frac{x}{R} \right) - \frac{x^2}{2R^2} \right]$$

+ Can solve using Lagrangian $T - V$, etc

- Gravity + Orbits

• Inverse Square Law Orbits

+ We looked at these in PHYS 3202

by saying one object is much more massive so it didn't move

+ Now we know that we can write the Lagrangian as

$$L = \frac{1}{2} m_1 \dot{\vec{r}}_1^2 + \frac{1}{2} m_2 \dot{\vec{r}}_2^2 + (Gm_1 m_2 / \vec{r}_1 - \vec{r}_2) \quad \left. \right\} m_1 m_2 = M$$

$$= \left(\frac{1}{2} M \vec{R}^2 \right) + \left(\frac{1}{2} \mu \vec{r}^2 + GM\mu/r \right)$$

+ So everything we did before still works for relative motion using fixed center of total mass M and orbiting object of reduced mass μ .

+ In our solar system $M_2 \gg M_1$ almost always, so $M \approx M_2, \mu \approx M_1$,
+ Kepler's 3rd law is slightly different for each orbiting object:

$$(T/2\pi)^2 \propto a^3/GM \text{ and } M = \text{total mass depends on } m_1, m_2 \text{ a little bit even if } m_2 \text{ is much bigger}$$

+ Each object orbits the CM in an ellipse.

If orbit semi-major axis is a , each has axis
 $a_1 = m_2 a/M, a_2 = m_1 a/M$

• Tidal friction

+ High tides are bulges of water due to moon's pull. B/c they rotate w/ earth. The pull back toward the moon slows earth's rotation

+ CM frame angular momentum $\vec{J} = M \vec{r} \times \vec{p} + I_{\text{Earth}} \vec{\omega}_{\text{Earth}}$
is conserved b/c external torques vanish on average

+ I_{Earth} very small, $M \vec{r} \times \vec{p} \approx 2\pi I_{\text{Earth}} \vec{\omega} = M \vec{r} \times \vec{v}$
by Kepler's 3rd, $I_{\text{Earth}} = I_{\text{Earth}} \omega_{\text{Earth}}$ where

$I_{\text{Earth}} = \text{earth moment of inertia}, \omega_{\text{Earth}} = \text{earth rotation frequency}$

+ As ω_{Earth} decreases, orbit axis a increases
Energy transfer to the moon's orbit!

• Restricted 3-body Problem

+ In general, it's impossible to solve for the motion of 3 objects interacting gravitationally except on a computer. This is the 3-body problem or N-body problem. Describes parts of our solar system, formation of structure in universe, etc.

+ In 3-body problem, objects + masses are primary m_1 , secondary m_2 , and tertiary m_3 with $m_1 > m_2 > m_3$

+ The restricted 3-body problem makes 3 assumptions

- Tertiary is very light $m_3 \ll m_1, m_2$, so its gravity does not affect the motion of m_1 or m_2 .
- The primary + secondary have circular orbits around their CM.
- All motion is in one plane.

+ Work in reference frame with primary/secondary CM rotating w/ the primary + secondary orbits. Then primary + secondary are at fixed positions $a_1\hat{x}, -a_2\hat{x}$.

The frame's angular velocity is $\omega\hat{z}$ where $\omega^2 = G(m_1+m_2)/(a_1+a_2)^3$

+ In this frame, the tertiary is at $\vec{r} = x\hat{x} + y\hat{y}$ by assumption. It experiences force (including fictitious)

$$\vec{F} = -m_3 \left(\frac{Gm_1}{r_1^3} \vec{r}_1 + \frac{Gm_2}{r_2^3} \vec{r}_2 \right) - 2m_3\vec{\omega} \times \vec{r} + m_3\omega^2 \vec{r}$$

where $\vec{r}_1 = \vec{r} - a_1\hat{x}$, $\vec{r}_2 = \vec{r} + a_2\hat{x}$ are relative displacements

+ There are 5 Lagrangian points where $\vec{F} = 0$

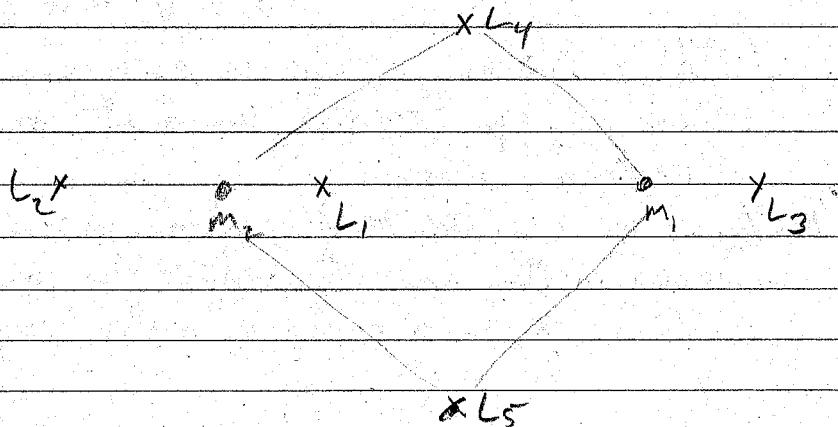
for stationary tertiary. These are on \hat{x} axis:

L_1 between $m_1 + m_2$; L_2 past m_2 ; L_3 past m_1 .

Motion here is unstable: tertiary will move away if

L_4 + L_5 are at vertices of equilateral triangle w/ m_1, m_2 .

Motion here is stable due to Coriolis force



- Rocket Propulsion

- A rocket ejects fuel at a fixed speed u relative to the rocket
 - + Ejection of fuel changes the mass of the rocket
 - + We have to account for total momentum of the ejected fuel in Newton's 2nd law
 - + At time t , the rocket (plus fuel inside) is $m(t)$
 - Ejected at rate \dot{m} fuel at rate $m = \dot{m}dt/dt < 0$.
 - Change in momentum in initial rocket rest frame.
 - + After Δt , after ejection of $\dot{m}\Delta t$ of fuel, rocket's momentum
- $$P_R = (m + \frac{\dot{m}}{u} \Delta t)(v\Delta t) = m v \Delta t.$$
- + After it leaves the rocket, the fuel has momentum
- $$P_F = (-\dot{m}\Delta t)(-u) = \dot{m}u \Delta t$$
- The sign on \dot{m} is b/c the ejected fuel gains mass.
- + The initial momentum in this frame $= 0$.
- so
- $$\frac{dP}{dt} = P_R + P_F = mv + \dot{m}u = F_{ext}$$
- Since u = relative velocity of rocket + fuel,
this is true in any frame
- + This is the Tsiolkovsky rocket equation

• Solutions

- + For a rocket in space $F_{ext} = 0 \Rightarrow v = -\frac{\dot{m}}{m} u$.
with u constant, $v_0 = \text{initial velocity}$, $m_0 = \text{initial mass}$.

$$v = v_0 + u \ln(\frac{m_0}{m})$$

If $m \geq m_{min}$, there is a mass trend.
That's why rockets are built in stages (to reduce m_{min})

- + You can see the reading above including external forces. It usually helps to assume $\dot{m} = -\alpha = \text{constant}$ to make progress.